AME: A Cross-Scale Constellation of CubeSats to Explore Magnetic Reconnection in the Solar–Terrestrial Relation

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A major subset of solar–terrestrial relations, responsible, in particular, for the driver of space weather phenomena, is the interaction between the Earth's magnetosphere and the solar wind. As one of the most important modes of the solar–wind–magnetosphere interaction, magnetic reconnection regulates the energy transport and energy release in the solar–terrestrial relation. *In situ* measurements in the near-Earth space are crucial for understanding magnetic reconnection. Past and existing spacecraft constellation missions mainly focus on the measurement of reconnection on plasma kinetic-scales. Resolving the macro-scale and cross-scale aspects of magnetic reconnection is necessary for accurate assessment and predictions of its role in the context of space weather. Here, we propose the AME (self-Adaptive Magnetic reconnection Explorer) mission consisting of a cross-scale constellation of 12+ CubeSats and one mother satellite. Each CubeSat is equipped with instruments to measure magnetic fields and thermal plasma particles. With multiple CubeSats, the AME constellation is intended to make simultaneous measurements at multiple scales, capable of exploring cross-scale plasma processes ranging from kinetic scale to macro scale.

Keywords: cross-scale, constellation, magnetic reconnection, solar-terrestrial relation, CubeSats, mother satellite, space weather

INTRODUCTION

Magnetic reconnection is a fundamental process of plasma transport and energy conversion, with applications in the near-Earth space, solar plasmas, astrophysical systems, as well as in the laboratory. Magnetic reconnection regulates the energy transfer in the solar–terrestrial connection. In the solar corona, magnetic reconnection explosively releases energy and sets off solar eruptions such as the coronal mass ejections and solar flares. In the Earth's magnetosphere, the dayside and night-side magnetic reconnection regulates the energy flow and energy release within the magnetosphere.

Intrinsically, magnetic reconnection is a multi-scale plasma process, involving global-scale, meso-scale, and kinetic-scale structure and processes. The "breaking" and "reconnection" of magnetic field lines occur in a limited region. The consequence and the impact of reconnection on the near-Earth space, however, are of a global nature. The Earth's magnetosphere is an ideal laboratory for *in situ* exploration of magnetic reconnection. Existing spacecraft constellation missions, such as Cluster and MMS, focused on measurements of magnetic reconnection on ion and electron kinetic scales. Cross-scale and macro-scale physics of reconnection, which is crucial for a precise characterization of its role in the context of space weather, remains to be explored. Generally, at least four spacecraft are needed to distinguish between spatial and temporal variations, and to measure gradients in the observed system. However, a large number of spacecraft is needed for multiple-scale measurements. While usage of standard spacecraft may imply a huge cost for a constellation mission, a CubeSat Constellation provides a cost-effective approach to achieve multiple-scale measurements.

With the above considerations, we propose a CubeSat Constellation "AME" (self-Adaptive Magnetic reconnection Explorer), which aims to make simultaneous cross-scale *in situ* measurements of space plasmas. In this paper, we describe the science rationale and the proposed mission profile of AME. The organization of the paper is as follows. The Magnetic Reconnection in Solar–Terrestrial Relation section presents an overview of magnetic reconnection in the context of space weather. The Scientific Rationale for Cross-Scale Constellation Missions section describes existing constellation missions and the need for future cross-scale constellation missions. The Self-adaptive Magnetic Reconnection Explorer section presents the science opportunity, mission profile, and payload of the AME mission. The Summary section gives the summary and conclusions.

MAGNETIC RECONNECTION IN SOLAR–TERRESTRIAL RELATION

A major part of solar–terrestrial relations, responsible in particular for space weather phenomena, is the transport of energy, mass, and momentum into the near-Earth space during the interaction between the solar wind and Earth's magnetosphere. One of the most important modes of the solar–wind–magnetosphere interactions is magnetic reconnection. Simply speaking, magnetic reconnection refers to the "cut" and "reconnection" of magnetic field lines in magnetized plasmas. This process is usually accompanied with energy conversion and transport of plasmas of different origins, responsible for explosive energy release phenomena such as solar flares, CMEs, and geomagnetic storms (e.g., [1]).

In the near-Earth space, magnetic reconnection regulates the energy entry and energy release within the magnetosphere. As illustrated in **Figure 1**, two primary sites for magnetic reconnection in the magnetosphere are the sub-solar point on the dayside and the rear magnetotail on the night-side. The dayside magnetopause reconnection enables the entry of magnetic flux and energy from the solar wind into the magnetosphere. This is the key process of the solar–terrestrial connection. The transfer of solar wind energy into the near-Earth space is a driver of geomagnetic activity. The night-side magnetic reconnection regulates the release of energy within the magnetosphere. The stretched magnetotail stores the energy from the solar wind. Magnetic reconnection releases the stored energy explosively within the magnetosphere and drives geomagnetic storms and substorms [3–5], producing various space weather effects such as enhanced relativistic electron fluxes in the radiation belts, increase in the ionospheric scale height, induced currents on ground power transmission lines, and bright auroras.

SCIENTIFIC RATIONALE FOR CROSS-SCALE CONSTELLATION MISSIONS

Magnetic reconnection is a spectacular example of a physical system dominated by multi-scale physics. In the large-scale region outside the site of where magnetic field lines "reconnect," the so-called MHD (magnetohydrodynamic) approximation applies. Plasma elements connected at one time by a single magnetic field line remain connected at subsequent times. The magnetic field lines appear to be "frozen-in" with the plasma motion. Magnetic reconnection is possible only where the MHD approximation breaks down, which happens in a thin non-MHD layer of electron-kinetic scales.

Understanding the nature of reconnection relies on the knowledge of both large-scale and kinetic-scale plasma properties. In the context of solar–terrestrial relation, one of the most important questions for reconnection is how much of the incoming solar wind plasma reconnects and flows into the magnetosphere, vs. how much of it flows around the magnetosphere without reconnecting [6]? This is substantially equivalent to the question of what determines the rate of magnetic flux transfer into the Earth's magnetosphere (reconnection rate). Whether the reconnection rate is determined primarily by the properties of the non-MHD layer in the reconnection site or, on the contrary, by the global dynamics and boundary conditions that govern the large-scale plasma flow is a question debated since the earliest studies of reconnection and still not conclusively settled. Another
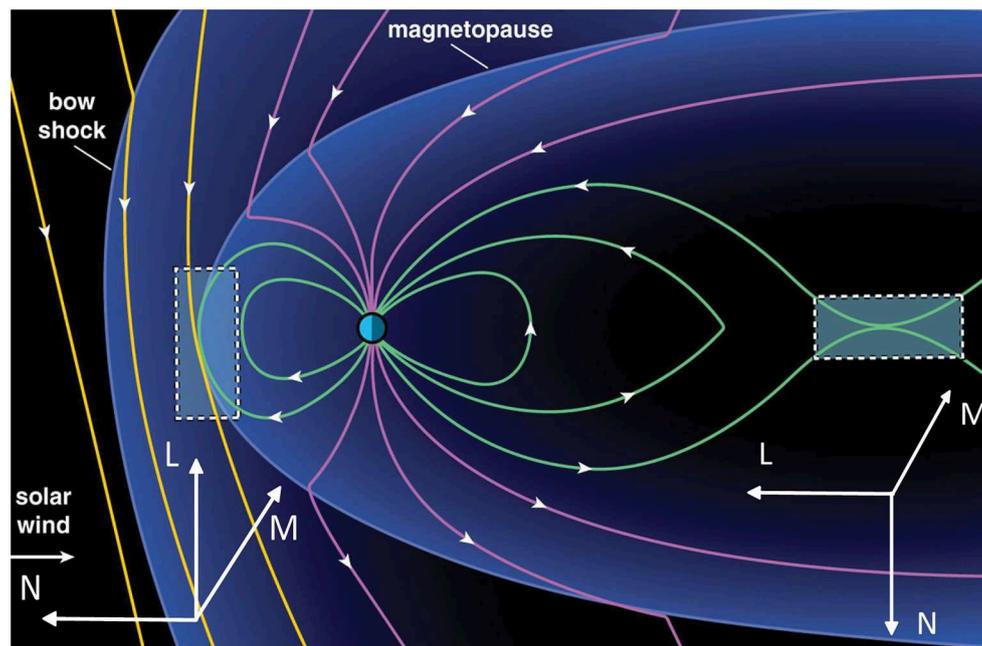
FIGURE 1 | Magnetic reconnection in the magnetosphere during southward magnetic field in the solar wind. Reconnection sites are indicated by two box areas. The LMN coordinate system represents the local normal boundary coordinate system for reconnection at different locations: L is the direction of the reconnecting magnetic field line, M is tangential to the normal and in the direction of the electric current, and N is the normal direction to the boundary layer. The figure is modified from Figure 1 of Burch et al. [2] under the terms of the Creative Commons Attribution 4.0 International License (http://creativecommons.org/licenses/by/4.0/).

important question of reconnection is what determines the reconnection onset. Since the initial theoretical studies of large-scale reconnection [e.g., [7]], it is expected that the onset and evolution of the reconnection process becomes governed not only by the physics of the diffusion region but also by the large-scale external magnetic fields and plasmas. The onset of reconnection must involve special conditions in thin current sheets—otherwise, it would occur all the time, which it does not. Similarly, the cessation of reconnection must involve interactions with a changing external environment; otherwise, it would continue forever. To date, we know neither what these conditions are nor where and when they occur, but it is exceedingly likely that their occurrence is shaped by the large-scale regions separated by the reconnecting layers.

In the near-Earth space, existing constellation missions such as Cluster, THEMIS, and MMS have fulfilled their science goals with four or five spacecraft placed at key positions or key separation distances from each other. The characteristic scales of previous and existing constellation mission in the magnetosphere are summarized in **Figure 2**. Cluster initially addressed plasma processes at the ion-scale and then measured two scales at the same time with two spacecraft separated from a few kilometers up to a few tens of thousands of kilometers. In such a case, however, the 3-D tetrahedron scale was lost with only four spacecraft. THEMIS addressed the storm and substorm process, and showed that magnetic reconnection in the near-Earth region was responsible for the initiation of magnetic energy release and the subsequent energy transport to the inner magnetosphere and ionosphere [3, 4]. The three inner THEMIS spacecraft

were measuring fluid scales (Earth radii), while the other two spacecraft were at 10 and 20 R_E away down the tail, covering the global scale. However, the five THEMIS probes were not in the form of a tetrahedron. Again, ion and electron scales could not be measured simultaneously, and the kinetic processes were not studied using multi-spacecraft. More recently, MMS focused on reconnection process at the electron scale with the four spacecraft in a tetrahedron shape and in the range of 7–50 km [2]. The electron diffusion region could be thus recognized and appropriately studied (e.g., [8–10]).

The next step forward to advance the physics of reconnection is, therefore, to measure simultaneously the electron-scale and ion-scale processes in the reconnection layer and their effect on the magnetosphere at fluid and global scales. It is timely to embark now on the questions of how the reconnection microphysics shapes the large-scale space environment and how the latter feeds back on the former. Answering this question requires a multi-scale observational approach.

A cross-scale mission allows putting magnetic reconnection process into the context of solar–terrestrial relation. Such a mission will answer how reconnection facilitates energy conversion on large scales and how this conversion couples to the dynamics of the small electron diffusion region. It can further address the critical question of why reconnection operates in small bursts at times and in large eruptions at others. It will furthermore shed light on the puzzle of whether large eruptions are a collection of small bursts occurring together, or whether they are a different, organized, type of reconnection altogether [11]. Understanding when and how these disruptions
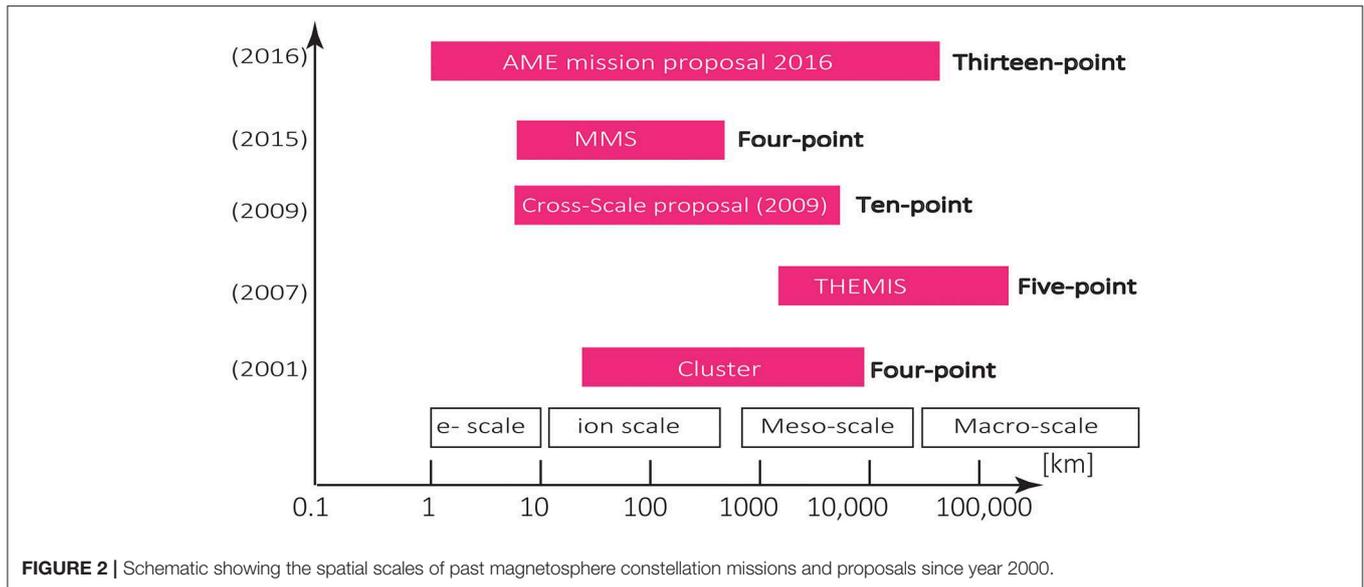
FIGURE 2 | Schematic showing the spatial scales of past magnetosphere constellation missions and proposals since year 2000.

occur is also critical to predict the harmful effects on the space environment, which are collectively referred to as "space weather" [12].

THE SELF-ADAPTIVE MAGNETIC RECONNECTION EXPLORER (AME) MISSION

Mission Overview

Previous and existing constellation missions of space plasma, for instance, Cluster [13], MMS [8, 14], and THEMIS [15], consist of four to five identical spacecraft. In principle, four spacecraft are needed to distinguish spatial and temporal variation at one specific scale. Such a constellation mission can provide detailed measurements at a single spatial scale. Owing to the limited number of spacecraft in previous constellations, however, the connection of pace plasma process across multiple scales cannot be fully explored.

The proposed self-Adaptive Magnetic reconnection Explorer (AME) mission consists of a base satellite and a constellation of 12 small CubeSats. The basic concept of the mission is to employ a constellation of spacecraft to take simultaneous measurements at differential spatial scales, exploring processes and structures from kinetic-scale to macro-scale in space plasmas. The AME mission proposal is currently in the stage of pre-phase A in the Strategic Priority Research Program on Space Science II, Chinese Academy of Sciences. In **Table 1**, we present a summary of the AME mission profile.

Scientific Objectives

The scientific rationale for cross-scale constellation missions has been well-described in the previous section. With the achievement of previous constellation mission at individual scale, many fundamental questions of magnetic reconnection remain to be explored. With all the above considerations in mind, we propose the following science objective for the cross-scale AME mission.

Science objective # 1. Understand the cross-scale connection of the multiple-scale processes of magnetic reconnection, ideally to the point of predicting when, where, and how fast magnetic reconnection occurs in the context of solar–terrestrial coupling.

TABLE 1 | AME mission profile overview.

AME mission summary

Main scientific objectives	1. Understand the cross-scale connection of the multiple-scale processes of magnetic reconnection, ideally to the point of predicting when, where and how fast magnetic reconnection occurs in the context of solar–terrestrial coupling 2. Provide understanding of cross-scale coupling in fundamental space plasma process such as in turbulence structures and waves
Payloads	CubeSats: Magnetometer, thermal ion/electron detector for each of the 12 CubeSats Mother satellite: Magnetometer, thermal ion/electron detector, high-frequency plasma wave analyzer, energetic particle detector, AC-electric field measurement
Spacecraft platform	CubeSats: a heritage from WN5000 Subsat Mother satellite: Platform has been designed, with a propulsion system and a Subsat-ejection mechanism
Mass budget	CubeSats: 12 × 67 kg Base satellite: 605 kg Dry mass: 1,409 kg Total mass with propellant: Option # 1, 2,084 kg; Option # 2, 2,495 kg
Mission duration	>4 years
Orbit and Launcher	Launcher: CA-3A Spacecraft mass load: ~2,200 kg Orbit, Option # 1: Phase 1, 1.2Re × 10Re; Phase 2, 1.2Re × 22 Re Option # 2: 10Re × 22Re orbit

The investigation of fundamental aspects of reconnection, such as when, where, and how fast it occurs, requires the understanding of the cross-scale coupling and the global aspect of reconnection. Answers to these questions are necessary for quantitative prediction of the role of reconnection in the context of space weather. Magnetic reconnection is usually studied in LMN (local-normal-boundary) coordinate system (see the description of LMN in **Figure 1**). In the following, we specify the science questions in this LMN coordinate.

Magnetic reconnection occurs in the thin current layer associated with magnetic field shear. However, magnetic field shear does not necessarily imply magnetic reconnection is occurring. Current layers without signatures of magnetic reconnection abound in space plasmas. Predicting when reconnection occurs relies on the understanding on the sequence of processes related to reconnection onset. **Figure 3** shows the observational signals related to the onset of magnetic reconnection. These signatures of reconnection are considered to be detected in the LN plane in the LMN coordinate system.

At the largest spatial and time scales, the upstream conditions are considered to affect the pre-condition of onset. The upstream plasma beta (the ratio of particle pressure to magnetic pressure), combined with the magnetic field shear angle, is believed to be an important parameter to determine if reconnection can occur for a certain current layer (e.g., [16]). On a smaller scale, the thinning of the current layer to an ion-scale is considered as a pre-onset signal in reconnection models. The thinning of the current sheet enhances the free energy for reconnection to occur. Thin current sheets can be unstable to a variety of plasma instabilities, including the lower-hybrid drift instability and the collisionless tearing mode instability (e.g., [17–19]). In addition, thin current layers down to the ion scale are also favorable for introducing two-fluid effects and fast reconnection (e.g., [20]). On ion scales to macro scale, the formation of diverging ion outflows from the reconnection site is the most definite signal for the occurrence of magnetic reconnection (e.g., [21, 22]). On the ion kinetic scale to the meso scale, the Hall fields and current, which have been also interpreted as an kinetic Alfven mode [23–25] or whistler mode [26], are considered as a key element for fast collisionless reconnection [20, 27]. On the electron kinetic scale, off-diagonal terms of the electron pressure tensor are expected to be related to the reconnection electric field [28, 29]. Similarly, off-diagonal terms of the ion pressure tensor are considered to balance the reconnection electric field on the ion-scale as well (e.g., [30–32]). Kinetic signals of reconnection at the electron-scale also include the crescent-like distributions, as was recently observed by the MMS mission (e.g., [8]). As described above, processes related to the reconnection onset occur at multiple scales. It would require a cross-scale constellation mission to resolve the relation and the time sequence of these processes. Such cross-scale measurements can be directly compared with particle-in-cell (PIC) and MHD-PIC simulation models (e.g., [33–35]).

In simple terms, the reconnection site where magnetic field lines "reconnect" takes the shape of an "X-line" along the direction of the electric current (in the M direction in LMN coordinate system). The extent of the "X-line" characterizes the spatial scale of reconnection in space. On the largest scale, a case study in the solar wind indicates that the X-line can be as large as several hundred Re [36]. Measurements in the magnetosphere suggest that the X-line of magnetopause reconnection may extend from several Re to tens of Re along the magnetopause [37, 38]. The magnetopause X-line may also appear patchy or of limited extent as inferred from *in situ* and ground-based measurements ([39] JGR; [40]). Compared with existing *in situ* multipoint missions, a cross-scale constellation would have much larger spatial coverage in terms of number of measurement points. As illustrated in **Figure 4**, such a constellation would allow systematic investigations of the spatial characteristics of the X-line.

A central question of magnetic reconnection is how fast it occurs [41, 42]. The meaning of "how fast reconnection occurs" includes at least two elements, the magnetic topology changes and the energy conversion. This is illustrated in **Figure 5**. In the scale of the reconnection diffusion region, the change in magnetic topology allows the mixture of plasmas from different origins. On the global-scale, magnetic flux transfer produced by magnetic reconnection is a crucial element in the global evolution of the magnetosphere [43]. Energy conversion occurs in a limited diffusion region in the Sweet–Parker model and also in the extended ion exhaust region downstream. In the Petschek model, the diffusion is limited, and energy conversion mostly occurs in switch-off shock at the boundary [42]. Recent observations suggest that coherent structures, for instance reconnection fronts and flux ropes, play an important role in energy conversion following reconnection onset [44, 45]. These coherent structures occur over a wide range of spatial scales. A cross-scale constellation mission can study the relation between different energy conversion processes at different scales.
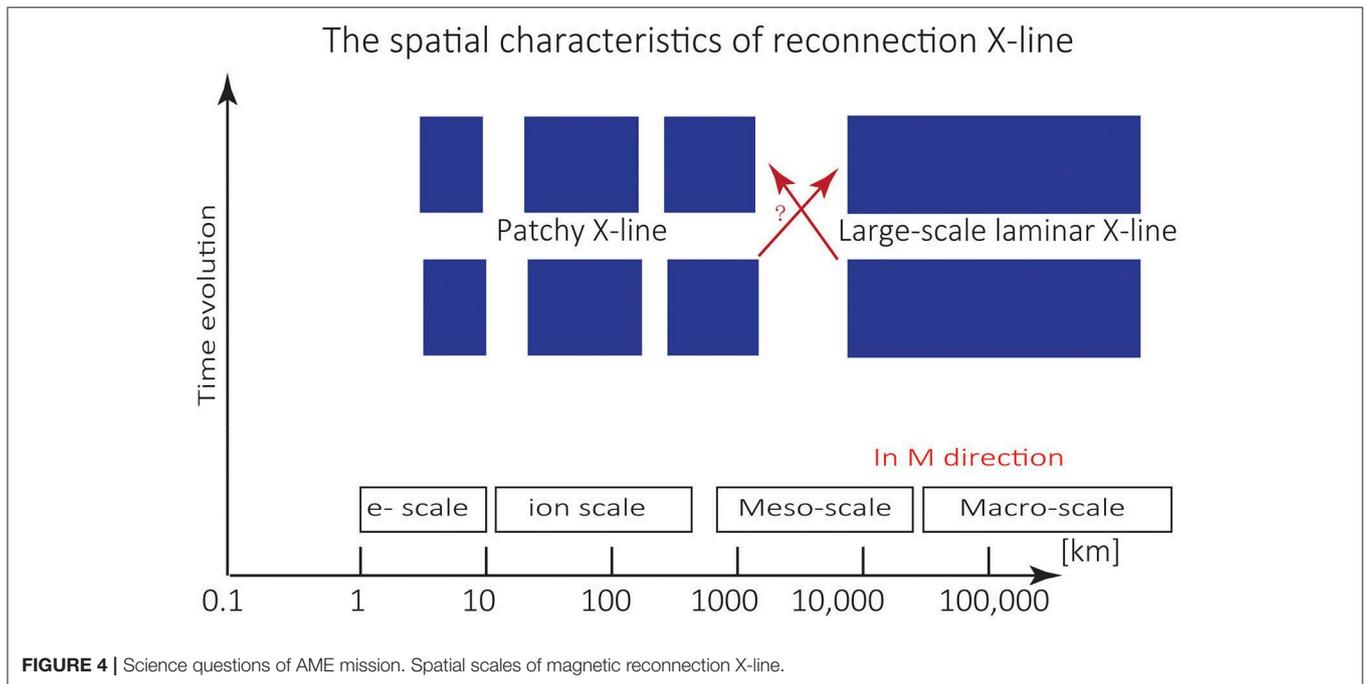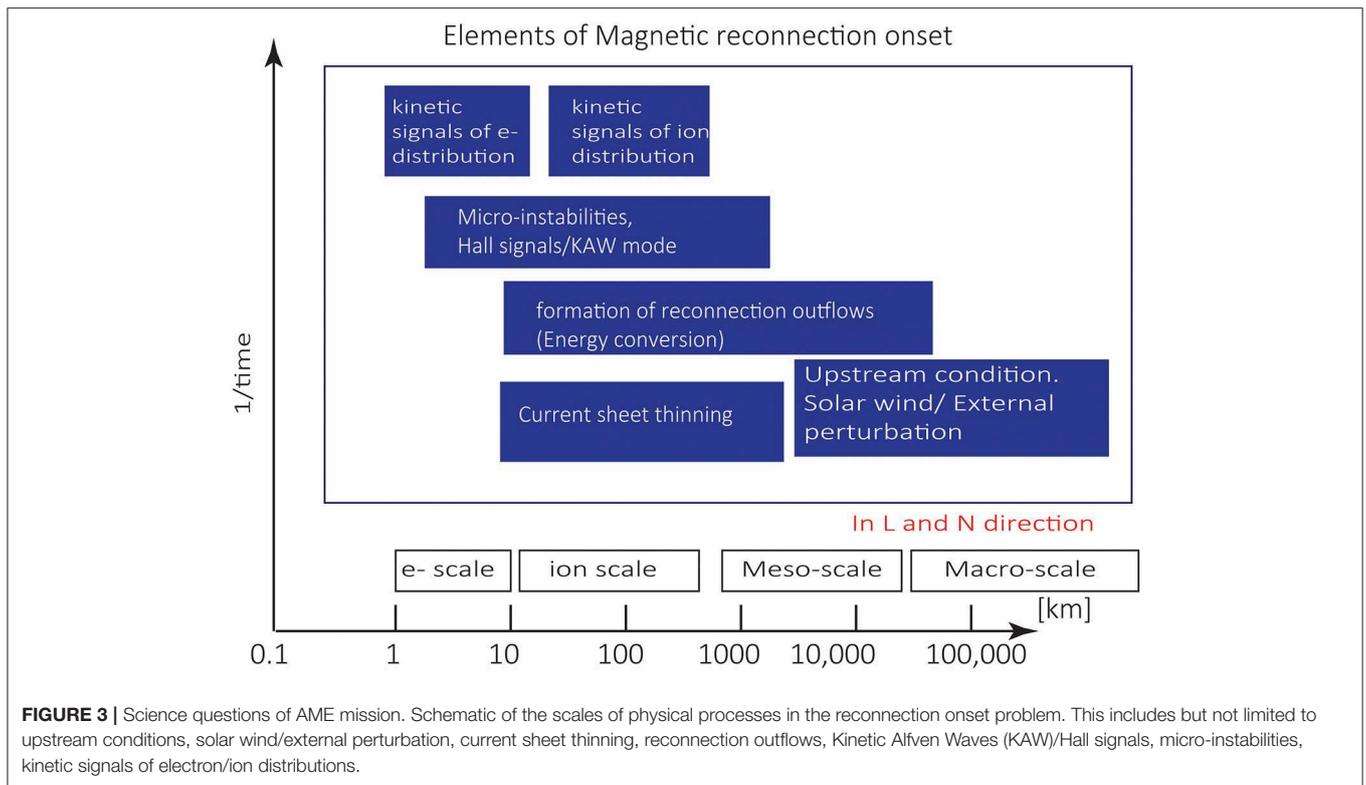
Science objective #2. Provide understanding of cross-scale coupling in fundamental space plasma process such as in turbulence, structures, and waves.

In addition to studying magnetic reconnection, a cross-scale constellation can provide multi-scale investigations on a variety of phenomena in space plasmas, including, but not limited to, turbulence, shock waves, current sheets, Kelvin Helmholtz instability, various types of discontinuities, flux ropes, dipolarization fronts, fast ion busty bulk flows, plasma waves, etc.

Turbulence and coherent structures universally exist in space plasmas. Kinetic-scale turbulence is strongly coupled to dissipative processes such as magnetic reconnection and shocks [46–48]. One form of turbulence in space plasmas is flow eddies. Recent MMS spacecraft provide detailed observation of electron-scale vortices [49, 50] and larger-scale flow vortices [51].

Current sheets (CSs) are widely spread in space plasmas and play an important role in storage and release of magnetic energy in the magnetosphere. Multiple-scale physics is a key ingredient of the formation of CSs. Observations and theoretical analysis revealed that multiple current–layer structures could form within a CS (e.g., [52–54]).

Kelvin–Helmholtz instability (KHI) often occurs on the Earth's magnetopause. Rolled-up K.-H. waves and the mixing of magnetosheath and magnetospheric were observed as Cluster

Elements of Magnetic reconnection onset

kinetic signals of e-distribution

kinetic signals of ion distribution

Micro-instabilities, Hall signals/KAW mode

formation of reconnection outflows (Energy conversion)

Upstream condition. Solar wind/ External perturbation

Current sheet thinning

In L and N direction

1/time

| e- scale | ion scale | Meso-scale | Macro-scale |

[km]

0.1   1   10   100   1000   10,000   100,000

FIGURE 3 | Science questions of AME mission. Schematic of the scales of physical processes in the reconnection onset problem. This includes but not limited to upstream conditions, solar wind/external perturbation, current sheet thinning, reconnection outflows, Kinetic Alfven Waves (KAW)/Hall signals, micro-instabilities, kinetic signals of electron/ion distributions.

The spatial characteristics of reconnection X-line

Time evolution

Patchy X-line   ?   Large-scale laminar X-line

In M direction

| e- scale | ion scale | Meso-scale | Macro-scale |

[km]

0.1   1   10   100   1000   10,000   100,000

FIGURE 4 | Science questions of AME mission. Spatial scales of magnetic reconnection X-line.

was skimming the magnetopause around the apogee (e.g., [55]). Such a large-scale structure as K.-H. waves would require cross-scale simultaneous measurements, with spacecraft at fluid scale, ion scale, and electron scale in order to be further understood (e.g., [56–58]).

The bursty bulk flow (BBF) in the magnetotail and the associated dipolarization front exhibit a multiple-scale nature, as revealed by multiple spacecraft measurements. At MHD scale, flow patterns of BBFs participate in the large-scale magnetosphere–ionosphere coupling process (e.g., [59, 60]). At

FIGURE 5 | Science questions of AME mission. Spatial scales of physical processes in the energy conversion and topology changes in reconnection.
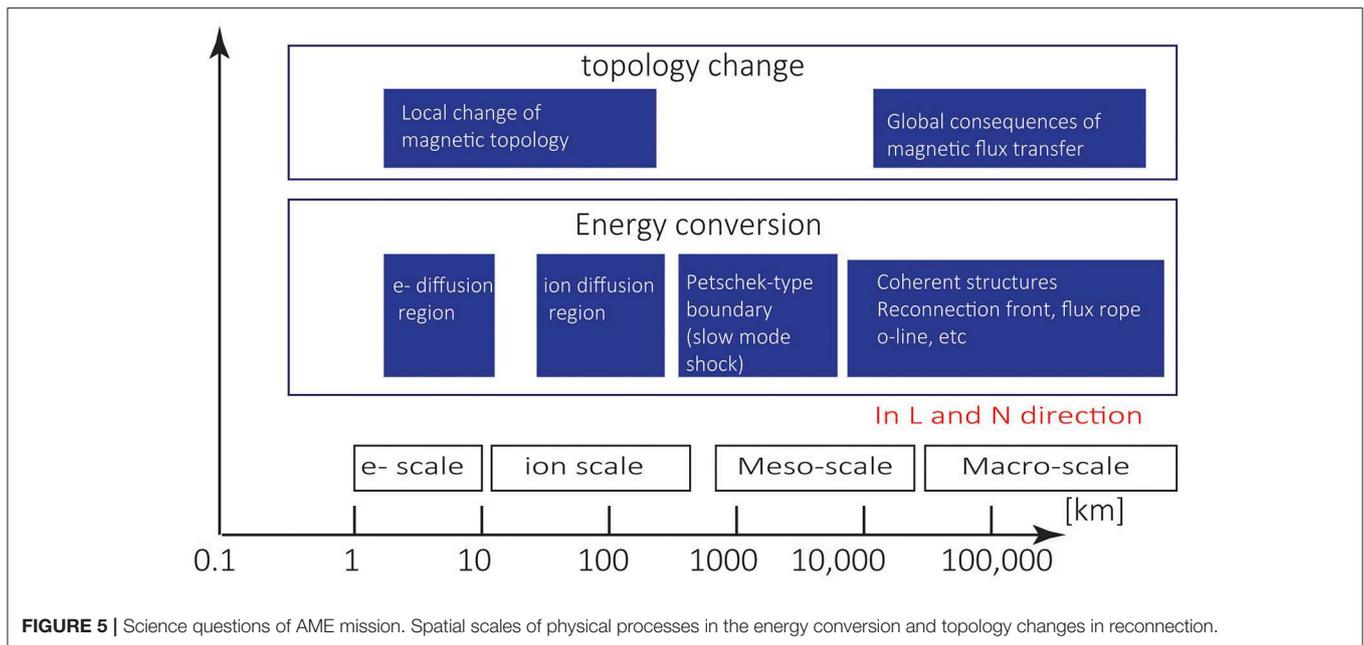
ion scale and electron scale, kinetic signatures of dipolarization fronts have been resolved by Cluster [61], THEMIS [62], and MMS (e.g., [63, 64]).

Plasma waves in the magnetosphere occur over a broad range of temporal and spatial scales. They can contribute to particle acceleration, scattering, and energy transport, which can modify magnetic reconnection and plasma boundaries. Plasma waves of a variety of scales are frequently observed in relation to reconnection, ranging from low-frequency kinetic Alfvén waves (KAW), to intermediate-frequency lower hybrid and whistler-mode waves, electrostatic broadband and solitary waves, as well as the high-frequency upper hybrid, Langmuir, and electron Bernstein waves [65].

Studies of structures and waves from four-point measurements usually rely on the assumption of linear interpolation and laminar structure at scales of spacecraft separation [66]. Studies of turbulence usually need to apply Taylor's hypothesis or equivalent assumptions to convert timescale to spatial scales. With a cross-scale constellation mission, some assumptions of the four-point/single-point analysis method can be relaxed or replaced. New multi-point analysis method, such as non-linear interpolation [67], may be applied to shed new insights in studies of turbulence, structures, and waves.

Scientific Payload

The magnetometer and thermal plasma (ion/electron) detectors are embarked on the main spacecraft and all the 12 CubeSats. In addition to these three payloads, a high-frequency plasma wave analyzer, an energetic particle detector, and a DC electric field instrument are employed on the main spacecraft.

The Magnetometer (MAG)

The vector magnetometer measures the three-dimensional magnetic field vector in the frequency range from DC to 40 Hz.

The configuration of MAG sensor and its deployable boom are shown in **Figure 6**. The MAG is a heritage from the payload for the CAS-ESA joint SMILE mission. Its specifications are shown in **Table 2**. The range of measurement for MAG is from −12,800 nT to +12,800 nT. The noise level is 0.1 nT (RMS). The mass of the MAG instrument, itself (without the deployable boom), is around 3.2 kg, and the boom is 5.7 kg. Its power is 5.0 W.

25 ev−30 keV Thermal Plasma (Ion/Electron) Detector

The thermal plasma (ion/electron) detector measures the three-dimensional particle distribution and its corresponding moments in the range of 5 eV−30 keV. The instrument includes two electrostatic analyzers, which take measurements for ions and electrons, respectively. The payload has a heritage from the light ion analyzer (LIA) for the CAS-ESA joint SMILE mission. The configuration of one electrostatic analyzer and electronic box is shown in **Figure 7**. The specifics of the plasma (ion/electron) detector are presented in **Table 3**. The energy resolution is 10%. The energy range is 5 eV−30 keV. The angle range of the detector is 360° in the azimuthal direction and 90° in the elevation direction. To make a measurement of the full 4Π solid space, either a spinning spacecraft or two identical sensors are needed. For a time cadence of 0.5 s, the data rate for each instrument is 48 kbps. The mass and power for each instrument is 3 kg and 6.0 W, respectively. The investigation to miniaturize the plasma detector is underway.

Electric Field Measurement

The double-probe electric field instrument uses the potential difference between two conducting spheres separated by booms to measure the electric field. The electric field is obtained by dividing the potential difference by the length between the probes. Since the main spacecraft is not spinning, and the length
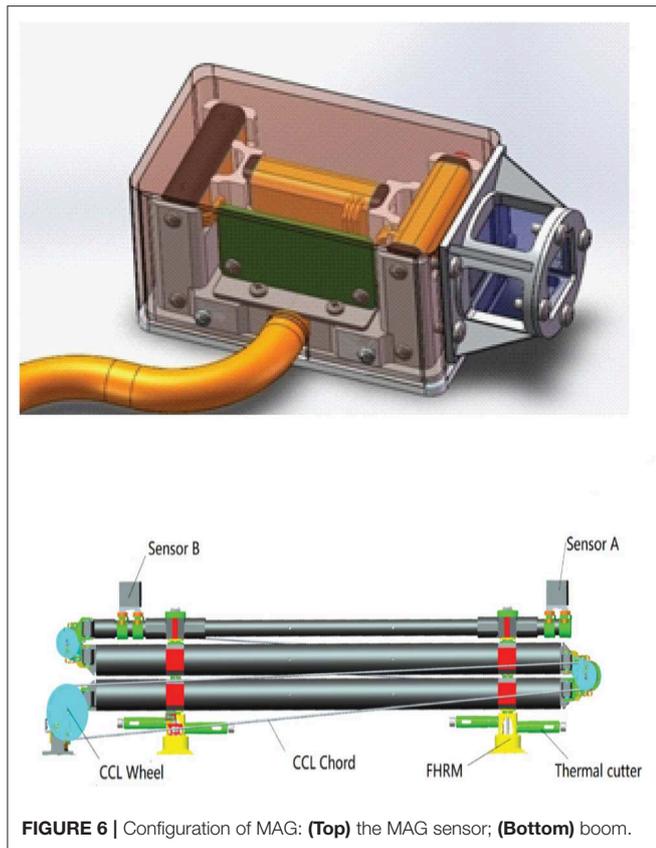
FIGURE 6 | Configuration of MAG: **(Top)** the MAG sensor; **(Bottom)** boom.

TABLE 2 | Summary of the performance of MAG.

Mass	0.2 kg per sensor (2x)
	2.2 kg E-box
	0.4 kg harness
	5.7 kg boom and release mechanism
Mass allocation	10.4 (+20%)
Power	5.0 W
Measurement range	±12,800 nT in orbit
	±64,000 nT for ground tests
Resolution	24 bit
Noise	0.1 nT (RMS)
Sample rate	40 Hz
Data rate	6 kbps

of the boom is limited (10 m for each boom), only the AC-electric field is measured as required. Spinning mother spacecraft and longer electric field boom will also be considered to implement accurate measurements of Dc and low-frequency electric fields. The specifications of the electric field measurement areas follow. The dynamic range of the electric field measurement is ±1,000 mV/m. The electric field resolution is 15 µV/m. The frequency band of the measurement is from 10 Hz to 5 kHz. The range of the voltage on the probe is −100 V∼ + 100 V. The resolution of the voltage on the probe is 3 mV. The mass and power is 1.5 kg
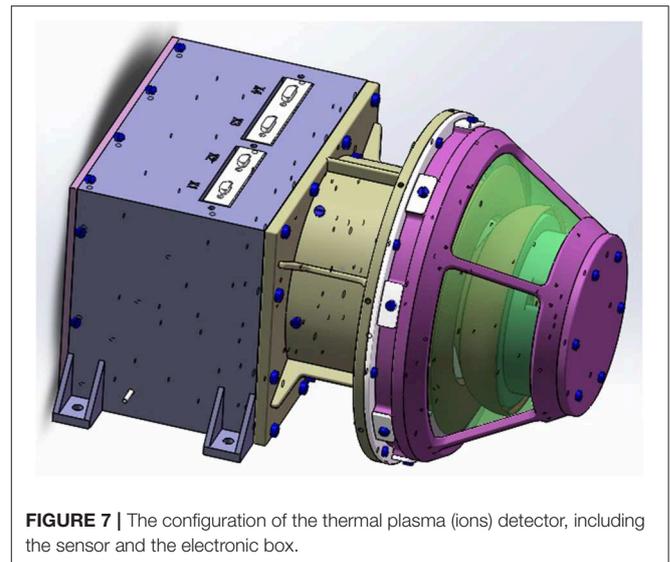
FIGURE 7 | The configuration of the thermal plasma (ions) detector, including the sensor and the electronic box.

TABLE 3 | Summary of the performance of thermal plasma (ion/electron) detector.

Measurement	Particle 3-D velocity distribution functions
Energy range	0.005–30 keV/q
Energy resolution ($\Delta E/E$)	≤0%
Angle range (azimuth)	360°
Angle range (elevation)	90°
Angular resolution (AZ × EL)	Azimuth: 30° coarse, 7.5° fine Elevation: ≤6°
Mass	3.0 kg
Power	6 W
Data rate	>48 kbps

and 2.5 W for each pair of the probes. The study of the possibility to employ a third pair of booms in the direction perpendicular to the spacecraft plane is underway.

High-Frequency Plasma Wave Analyzer

The high-frequency plasma wave analyzer employ a tri-axial search-coil magnetometer (SCM) to measure the 3D wave magnetic field vector in the frequency range from 10 Hz to 10 kHz. The noise level is of the designed SCM and is $<5 \times 10^{-3}$ nT/$\sqrt{}$Hz at 10 Hz, 2.5×10^{-4} nT/$\sqrt{}$Hz at 200 Hz, 5×10^{-5} nT/$\sqrt{}$Hz at 2 kHz, 6×10^{-5} nT/$\sqrt{}$Hz at 10 KHz. The waveform provided by the SCM, together with electric field data provided by the double-probe measurement, is processed inside the digital signal processor of the high-frequency plasma wave analyzer. A deployable boom would be necessary to avoid perturbations from the spacecraft on the measurements.

Energetic Particle Measurement

The energetic particle detector measures the energy spectra and the pitch angle of the energetic protons and electrons. The energetic particle detector includes three sets of sensors and one electronic box. Each set of sensors include one proton sensor
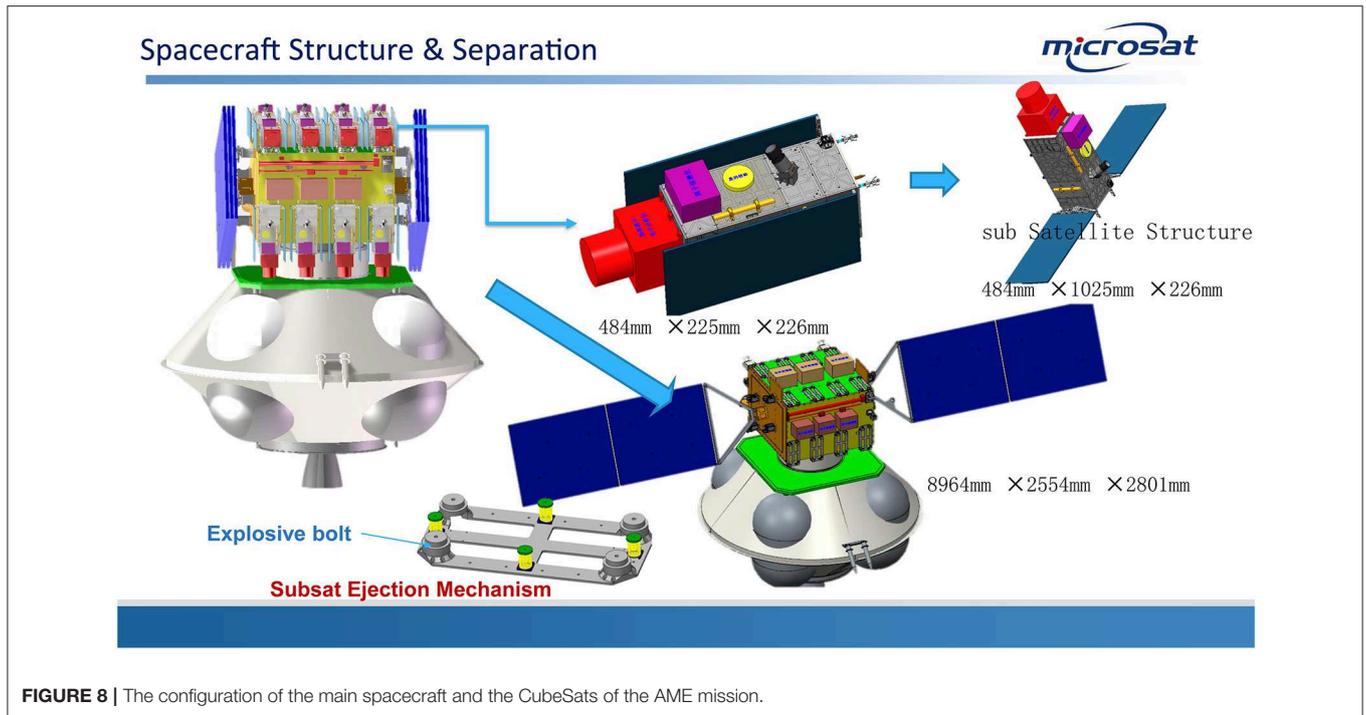
FIGURE 8 | The configuration of the main spacecraft and the CubeSats of the AME mission.
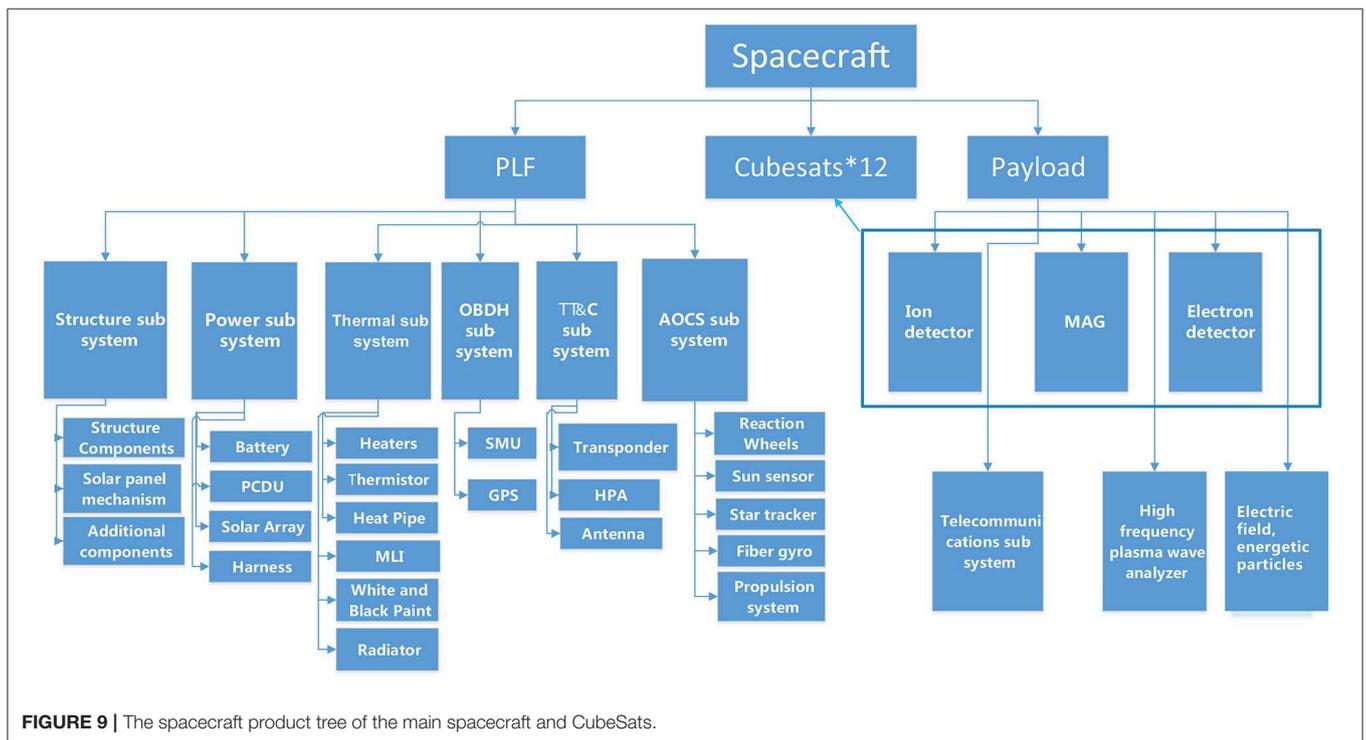

FIGURE 9 | The spacecraft product tree of the main spacecraft and CubeSats.

and one electron sensor. The energy range of the measured proton is 50 keV−4 MeV, the field-of-view is 40°, and the energy resolution is 20%. The energy range of the electrons is 50 keV−400 keV, the field-of-view is 30°, and the energy resolution is 20%.

Spacecraft, Orbit, and Launcher

The spacecraft configuration is shown in **Figure 8**. The left side shows the configuration of the combination of the AME constellation fleet. The 12 CubeSats are installed at the top of the main spacecraft. A release mechanism will deploy each CubeSat
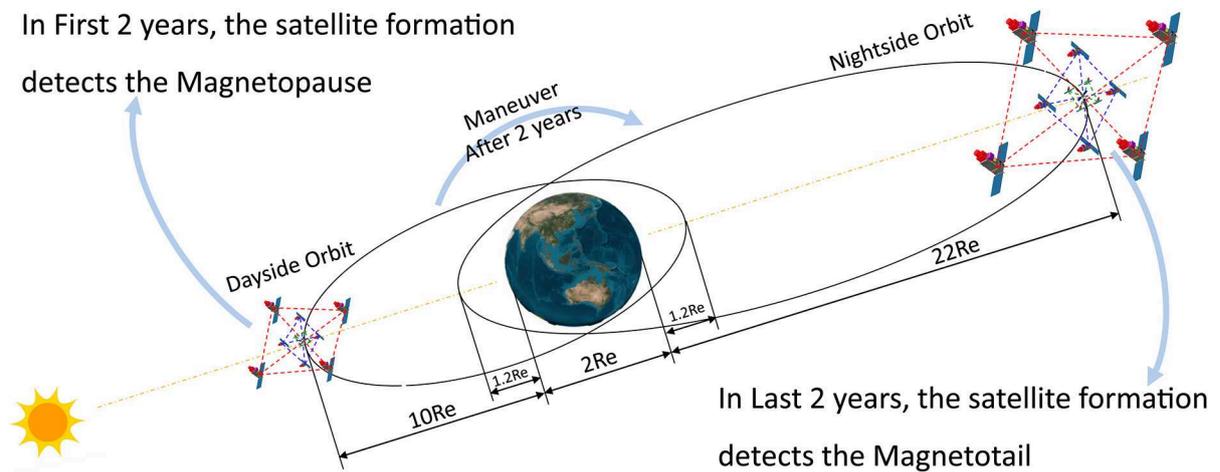
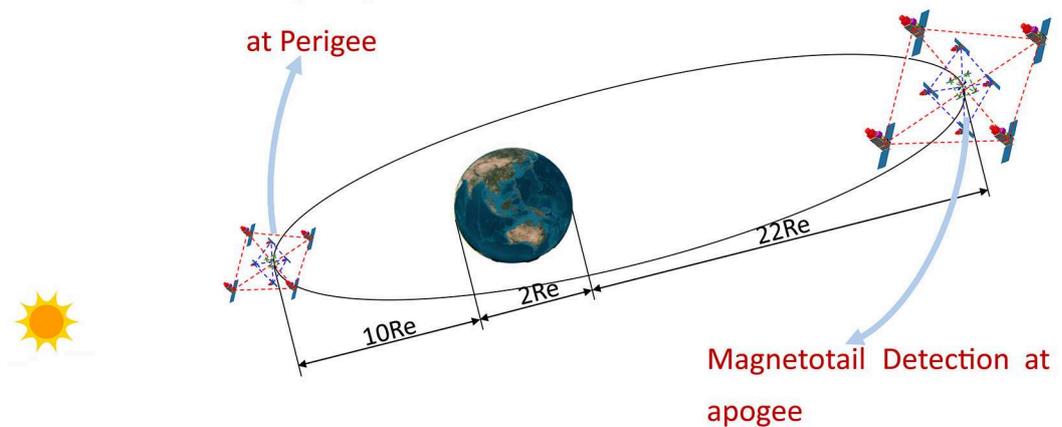**FIGURE 10 |** The configuration of the spacecraft orbit in option # 1.

FIGURE 11 | The configuration of the spacecraft orbit in option # 2.

when the spacecraft reaches the nominal orbit. The design of the CubeSats is shown in the middle top of the figure. The dimension of each CubeSat before (after) the deployment is 484 mm × 225 mm × 226 mm (484 mm × 1,025 mm × 226 mm). The main

spacecraft dimensions, after unfolding solar panels, are 8,964 mm × 2,554 mm × 2,801 mm.

The spacecraft product tree is presented in **Figure 9**. The main spacecraft includes the Platform (PLF) consisting

No.	Module	Subsystem	Mass at launch/kg	
			Mass Allocation	Total Mass
1	PF	EPS	58.0	529.0
2		OBDH	10.0	
3		TT&C	10.0	
4		AOCS	35	
5		Propulsion	318.0	
6		Structure	90.0	
7		Thermal	8.0	
8	PLM	--	76.0	76.0
9	SubSat	--	67.0(x12)	804.0
10	Propellant Of Scheme 1			675.0
11	Propellant Of Scheme 2			1086.0
12	Dry Mass			1409.0
13	Launch Total Of Scheme 1			2084.0
	Launch Total Of Scheme 2			2495.0

FIGURE 12 | The mass budget of the AME mission.

of six sub-systems. The structure sub-system includes the structure components, the solar panel mechanism and additional components. The power sub-system includes the batteries, the power control unit (PCDU), and the solar array and harness. The thermal sub-system includes the heaters, thermistor, heat pipe, multi-layer insulation (MLI), white and black paint, and radiator. The onboard data handling (OBDH) sub-system include the global position system (GPS) and solid state memory unit (SMU). The telemetry, tracking, and commanding (TT&C) sub-subsystem includes the transponder, high-power amplifier (HPA), and antenna. The TT&C system is responsible for uploading the telemetry commands, downloading all telemetry data, and ranging and tracking of the satellite. We rely on the inter-satellite communication system to transmit science data from CubeSats to the mother satellite. All the data are then transmitted from the mother satellite to ground stations via a high-gain antenna and a Ka band transmitter. The TM and TC data rate are 8,192 and 2,000 bps, respectively. For the inter-satellite communication system, a broad-beam UHF antenna and a UHF transmitter are essential. The main specifications for the UHF antenna are frequency, 436 MHz; bandwidths, $\Delta f \geq$ 2 MHz; gain, $G \geq -3$ dBi (within Beam range); polarization mode, linear. The specifications for the UHF transmitter are frequency, 436 MHz; output power, >6 W; data rate, 4 kbps; carrier modulation, OQPSK; data coding, differential coding of the NRZ-L type; channel coding, convolutional coding (7, 1/2), of constraint length 7 and ratio 1/2. The configuration of UHF antenna and transmitter is feasible and has been widely used for inter-satellite communication missions, for instance, xx-6 and Kuai Xiang mission at the Innovation Academy for Microsatellites. The Attitude and Orbit Control System (AOCS) sub-system includes the reaction wheels, the sun sensor, star tracker, Fiber Gyro, and propulsion system. The average power for the platform and payload of the main satellite is 200 and

180 W, respectively. The average power for the platform and payload of each CubeSat satellite is 67 and 30 W, respectively.

Two of the spacecraft orbits are envisaged to fulfill the science requirement to measure the reconnection near the dayside magnetopause and in the night-side magnetotail. The orbit option # 1 is shown in **Figure 10**.

Option #1 includes a dayside phase of 2 years and a night-side phase of 2 years. In the first dayside phase of the mission, the perigee is 1.2Re, the apogee is 10Re in the dayside, and the inclination is 23.5°. In the night-side phase, the perigee is 1.2Re, the apogee is 22Re in the night-side, and the inclination is 23.5°. Option # 2 is an optional orbit that can measure the magnetopause reconnection near the perigee and magnetotail reconnection at the apogee. Option #2 is shown in **Figure 11**. The parameter of the orbit is as follows. The perigee is 10Re, the apogee is 22Re, and the inclination is 23.5°.

The main satellite and the 12 CubeSats form multiple tetrahedrons around the main satellite. The baseline option of the satellite formation (formation #1) is to divide the 12 CubeSats into three four-satellite groups. Each four-satellite group forms a tetrahedron at a specific scale. The three tetrahedra have three different scales, at 1–10 km (electron scale), 100–1,000 km (ion scale), and 1–3Re (macro scale). Another optional formation (formation #2) is to divide the 12 CubeSats into four three-satellite groups. Each group of three CubeSats, together with the main spacecraft, forms a tetrahedron. The fleet forms four tetrahedron at 1–10 km (electrons scale), 100 km (ion scale), 1,000–10,000 km (meso scale), and 3–5Re (macro scale). In either formation, the configuration of each tetrahedron is self-adjustable to form a suitable configuration in the LMN coordinate system. The input for adjusting the configuration is based on either real-time measurement data or advance planning. The propulsion system of the CubeSats will have sufficient fuel to maintain and adjust the tetrahedral formation.

The mass budget of the mission is shown in **Figure 12**. The platform (PF) part of the main satellite is 529 kg. The payload (PLM) part of the main satellite is 76 kg. The mass of each CubeSat is 67 kg, including 26 kg of payload and 9.5 kg of propulsion system. The total dry mass for the main satellite and the 12 CubeSats is 1,409 kg. The Launcher is Long March CZ-3A, which has the capacity to carry a spacecraft mass of 2,200 kg with a maximum envelope of 3,350 * 8,887 mm. Including the propellant, the total mass for launch is 2,084 kg for option # 1 and 2,495 kg for option # 2. The launch site is expected to be at Xichang. The ground stations are Sanya and Miyun.

SUMMARY

A cross-scale constellation mission allows investigating the global-scale consequences of magnetic reconnection in the near-Earth space and its coupling to the micro-scale physics in the reconnection layer. Such a constellation mission makes it possible to put the investigation of magnetic reconnection into the context of solar–terrestrial relations and space weather.

A constellation of four spacecraft is needed to make measurements at one specific scale. A multiple-scale constellation thus requires a large number of spacecraft. To provide a cost-effective approach to achieve such multiple-scale measurements, we propose the AME Constellation consisting of 12 small CubeSats surrounding a base satellite. This constellation enables simultaneous space plasma measurements at different spatial scales ranging from kinetic-scale to macro-scale. In this paper, we have described the scientific rationale and the profile of the proposed AME mission. This is a logical next step forward to advance the understanding of reconnection in the context of solar–terrestrial relations.

DATA AVAILABILITY STATEMENT

All datasets generated for this study are included in the article/supplementary material.

AUTHOR CONTRIBUTIONS

All authors participated on the discussions of the results and commented on the paper. LD, CW, PE, VA, MH, WG, VV, ZC, DG, HF, RN, MD, and QL contributed to the writing and editing of the paper.

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Conflict of Interest: The authors declare that the research was conducted in the absence of any commercial or financial relationships that could be construed as a potential conflict of interest.

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